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6 **Discrete and Continuous Models for Static and Modal Analysis of Out of Plane**  
7 **Loaded Masonry**

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23 **Abstract**

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25 A critical review of analytical and numerical models for studying masonry out of  
26 plane behaviour is presented. One leaf historical masonry, composed by rigid blocks  
27 arranged regularly with dry or mortar joints, is considered. Discrete model with rigid  
28 blocks, Love-Kirchhoff and Reissner-Mindlin plate models and 3D heterogeneous  
29 FEM are adopted. An existing simple and effective discrete model is adopted and  
30 improved by applying matrix structural analysis techniques for static and modal  
31 analysis of masonry walls in the elastic field, but the formulation allows to account  
32 for material nonlinearity. Elastic parameters of both plate models are based on an  
33 existing compatible identification between 3D discrete model and 2D plate models.  
34 Static and modal analysis of masonry walls with several boundary conditions are  
35 carried on, numerical tests account for in plane size of heterogeneity and structure  
36 thickness by means of in and out of plane scale factors. Results show that discrete  
37 model is simple and effective for representing masonry behaviour, especially when  
38 size of heterogeneity is smaller than overall panel size. Decreasing in plane scale  
39 factor, plate models converge to the discrete one, but the Reissner-Mindlin one  
40 shows a better convergence and also allows adopting a simple FE for performing  
41 numerical analysis.  
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44 **Keywords:** masonry, out of plane analysis, modal analysis, discrete model,  
45 Kirchhoff plate, Mindlin plate, heterogeneous material, homogenisation, elasticity.  
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# 1 Introduction

Masonry is a structural material obtained by composition of natural or artificial blocks connected by dry or mortar joints. For this type of material, size of heterogeneity (or size of block) is often not negligible with respect to the global size of a structural element; then, several ad-hoc models have been developed in the last decades adopting different approaches.

The model that may appear as the more simple one among others for representing masonry behaviour is a heterogeneous Finite Element (FE) model. The first example of such a model type was limited to the in plane case, characterised by blocks modelled with FE quadrilateral elements and joints modelled with one-dimensional elements [1,2]. Small improvements of these initial works were performed by Tzamtzis & co-workers by defining three-dimensional elements and joints, but limiting the field of analysis to the in plane case [3,4]. However, the limits of this approach are represented by the large number of degrees of freedom involved and the consequent computational effort for the analysis of macro-scale problems.

Another class of numerical models frequently adopted for representing masonry behaviour is discrete modelling, that is characterised by rigid or deformable elements in contact, in order to represent natural or artificial blocks in contact by means of dry or mortar joints. Into this class of models, the discrete element method (DEM) is one of the most representative approaches. Such a method is characterised by distinct elements, with finite size and independent degrees of freedom, that can be subject to finite displacements and rotations; moreover, contacts between elements can vary during analysis and are automatically recognized by the model. DEM had been introduced by the pioneering works of Cundall in the field of rock mechanics [5,6], that started considering the plane case and created the well-known program UDEC [7], and continued modelling three-dimensional problems [8,9], creating the program 3DEC [10]. Several developments in the DEM field are represented by the combination of discrete and finite elements [11,12] accounting for block deformability; however these models are still limited to in plane analysis. Another type of discrete models is represented by those adopted in Discrete Deformation Analysis (DDA, [13]) that are characterised by deformable blocks by means of uniform strain and stresses in plane state. This model was extended to the 3D case even if at preliminary stage [14,15].

The discrete models cited above were created for modelling granular materials and for studying rock mechanics, in several cases such models were extended to the field of masonry structures with results comparable with those obtainable with other classes of models; the work of Lemos [16] presents a deep review of DEM applied to masonry structures, furthermore the recent book edited by Bagi, Sarhosis and Milani [17] collects an up-to-date review of DEM for masonry and other discrete approaches.

However, historical masonry is frequently made of strong and rigid natural or artificial blocks and weak, thin and deformable mortar joints. For this reason, numerical models, characterised by rigid blocks, with deformability concentrated at mortar joints or dry contacts, subject to small displacements that do not vary contact topology, should be sufficiently accurate and effective. Effectiveness is given by the

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124 small number of degrees of freedom involved in the analysis that allows to model  
125 structures starting from small masonry panels to building facades and bridges. It is  
126 worth noting that these models cannot be defined DEMs but still remain discrete  
127 models. Considering the in plane case, this type of model was adopted by many  
128 authors in linear and non-linear fields [18-22]. In particular, the discrete and  
129 heterogeneous models introduced by Cecchi and Sab [20] were extended to the out  
130 of plane case by also performing homogenisation procedures [23-25]. Similarly, the  
131 'rigid-body-spring-model' introduced by Casolo was effectively extended to the out  
132 of plane case, in linear and nonlinear fields [26,27] and it was also compared with a  
133 homogenised model [28].

134 Heterogeneous and discrete models are often linked with continuous materials  
135 equivalent to masonry, obtained by means of identification or homogenisation  
136 procedures. Continuous models are another class of models that are generally  
137 adopted for studying masonry behaviour at macroscale level, when both  
138 heterogeneous and discrete models start to be inapplicable due to the huge number  
139 of degrees of freedom involved in the analysis of masonry buildings. Considering  
140 the in plane case, standard Cauchy models were obtained applying periodic  
141 homogenisation techniques and considering the elastic behaviour of both brick and  
142 mortar [20,24,29]; moreover, micropolar or higher order continua were taken in  
143 consideration [30-35]. Considering the out of plane case, Stefanou et al. [36]  
144 performed a 3D Cosserat homogenisation of regular masonry composed by rigid  
145 elements in the linear elastic field and then proposed a FE formulation for Cosserat  
146 elastic plates [37]. However more generally, research in the 3D or out of plane field  
147 has been generally focused on nonlinear masonry behaviour for performing  
148 nonlinear, limit and stability analysis of walls, facades and buildings [27,38-46].  
149 Recently, Ferreira et al. [47] presented an accurate literature review related to the  
150 analysis of unreinforced masonry out of plane loaded.

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152 Considering the field of analysis based on homogenisation or identification  
153 procedures, plate models are often adopted for modelling out of plane masonry  
154 behaviour. In particular, the already cited works of Cecchi and Sab [23,24] show an  
155 identification procedure that is based on the balance of internal work in the discrete  
156 model and in the continuous one for a class of regular motions. In this field, an  
157 important problem is represented by how kinematic, dynamic and constitutive  
158 prescriptions of a discrete system are transferred to the continuous one. Hence,  
159 constitutive functions of the plate model may be different. For example, a Love-  
160 Kirchhoff plate model was proposed by Cecchi and Sab [20] for the case of both  
161 rigid and deformable blocks by means of homogenisation procedures, whereas  
162 Cecchi and Sab [24] studied both Love-Kirchhoff and Reissner-Mindlin plate  
163 models for rigid blocks connected by elastic interfaces by means of a 3D discrete  
164 model and homogenisation procedures.

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166 Recently, Baraldi et al. [48] have presented a review of several numerical models,  
167 heterogeneous, discrete and continuous, that may be adopted for modelling the  
168 mechanical behaviour of masonry, with particular attention to out of plane loaded  
169 panels having a specific regular texture. The present work aims to extend the initial  
170 review by adding further information about out of plane displacement and rotation  
171 fields obtained with linear static analysis. Moreover, this work aims to extend the  
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178 campaign of numerical tests to the field of modal analysis, by means of a simple and  
 179 effective approach for studying the discrete system, based on the determination of  
 180 the stiffness matrix of the masonry assemblage. In addition, analytic solutions  
 181 relative to natural frequencies of homogenised plates simply supported along edges  
 182 are presented.

183 Hence in this work, numerical evaluation of the differences between a discrete  
 184 model with rigid blocks, heterogeneous FEM and homogenised plate models is  
 185 carried on for several case studies, performing static and modal analysis and  
 186 considering several boundary conditions. The effect of varying in plane  
 187 heterogeneity size (block width respect to panel width) is considered, as it has been  
 188 done for the in plane case by several authors [32,35,49]; moreover, the effect of  
 189 block aspect ratio (block width with respect to block height) and out of plane scale  
 190 factor (block thickness with respect to panel width) are taken into account.

191 In order to represent the behaviour of historical masonry, characterised by block  
 192 stiffness larger than mortar stiffness and joint thickness smaller with respect to block  
 193 size, the discrete model adopted in this work and the corresponding homogenised  
 194 plate models are based on the following hypotheses: i) masonry structure composed  
 195 by infinitely rigid blocks subject to small displacements and with fixed contact  
 196 topology, ii) mortar joints modelled as elastic interfaces. It is worth noting that the  
 197 elastic behaviour considered may not be correct for studying masonry structures,  
 198 given that such structures present a strong nonlinear behaviour even at low stress  
 199 levels; however, the proposed review represents an initial step for performing  
 200 numerical tests of out of plane loaded panels, that can be extended to the nonlinear  
 201 field in further developments of this work, adopting, for example, a Mohr-Coulomb  
 202 yield criterion for interface actions, following the numerical tests recently dealt with  
 203 by authors both for the in and out of plane cases [50,51]. Moreover, the proposed  
 204 campaign of modal analyses will allow to perform structural identification of  
 205 masonry specimens by comparing numerical results with laboratory or in situ tests.

## 213 2 Discrete model

214 This work considers a regular and periodic masonry assemblage, characterised by  
 215 equal rigid blocks arranged regularly with aligned horizontal joints and vertical  
 216 joints staggered by block half width. This model is defined as Discrete Rigid Block  
 217 Model (DRBM). A representative elementary volume (REV) is considered (Figure  
 218 1), characterised by a generic block  $B_{i,j}$  surrounded by six blocks by means of six  
 219 interfaces or joints  $S_{k_1,k_2}$ , with  $k_1, k_2 = \pm 1$  for horizontal interfaces and  $k_1 = \pm 2, k_2 =$   
 220  $0$ , for vertical interfaces (Figure 1). Block dimensions are:  $a$  (height),  $b$  (width) and  $s$   
 221 (thickness). It is worth noting that this contact topology is assumed to be fixed  
 222 during the analysis. Considering rigid block and small displacements hypothesis, the  
 223 displacement of a generic block is represented by a rigid body motion referred to the  
 224 motion of its centre and the rotation with respect to its centre:

$$228 \mathbf{u}^{i,j}(\mathbf{y}) = \mathbf{u}^{i,j} + \boldsymbol{\Omega}^{i,j}(\mathbf{y} - \mathbf{y}^{i,j}) \quad (1)$$

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where  $\mathbf{y}^{i,j}$  is the position of block centre in the Euclidean space and considering the out of plane case,  $\mathbf{u}^{i,j} = u_3^{i,j}$  is the out of plane translation of block centre and  $\mathbf{\Omega}^{i,j}$  is the rotation skew tensor collecting block rotations with respect to  $y_1$  and  $y_2$  axes:

$$\mathbf{\Omega}^{i,j} = \begin{bmatrix} 0 & 0 & \omega_2^{i,j} \\ 0 & 0 & -\omega_1^{i,j} \\ -\omega_2^{i,j} & \omega_1^{i,j} & 0 \end{bmatrix} \quad (2)$$

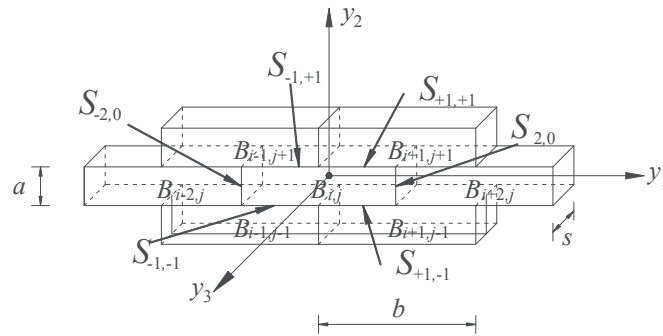


Figure 1. Discrete rigid block model (DRBM), running bond Representative Elementary Volume (REV)

Out of plane translation and rotations may be collected in  $\mathbf{q}^{i,j} = \{u_3^{i,j} \ \omega_1^{i,j} \ \omega_2^{i,j}\}^T$ . Following the procedure described by Cecchi and Sab [24] for the out of plane case, the interactions between two adjacent blocks  $B_{i,j}$  and  $B_{i+k_1,j+k_2}$  through a generic interface  $S_{k_1,k_2}$  are represented by unknown distribution of stresses, normal and tangential,  $\sigma^{k_1,k_2} = \{\sigma_n^{k_1,k_2} \ \sigma_t^{k_1,k_2}\}^T$ , with normal stress assumed positive in case of tension and negative in case of compression. Integrating stresses over the interface area, interface force  $f_3^{k_1,k_2}$  and couples  $c_1^{k_1,k_2}, c_2^{k_1,k_2}$  are obtained and collected in  $\mathbf{f}^{k_1,k_2} = \{f_3^{k_1,k_2} \ c_1^{k_1,k_2} \ c_2^{k_1,k_2}\}^T$ . Such stresses and interactions are related to the relative displacement and rotations between adjacent blocks, that are defined by:

$$\begin{aligned} d_3^{k_1,k_2} &= u_3^{i+k_1,j+k_2} - u_3^{i,j} + k_1 \frac{b}{2} \frac{(\omega_2^{i+k_1,j+k_2} + \omega_2^{i,j})}{2} - k_2 a \frac{(\omega_1^{i+k_1,j+k_2} + \omega_1^{i,j})}{2}, \\ \delta_1^{k_1,k_2} &= \omega_1^{i+k_1,j+k_2} - \omega_1^{i,j}, \\ \delta_2^{k_1,k_2} &= \omega_2^{i+k_1,j+k_2} - \omega_2^{i,j}, \end{aligned} \quad (3a-c)$$

and that may be collected in  $\mathbf{d}^{k_1,k_2} = \{d_3^{k_1,k_2} \ \delta_2^{k_1,k_2} \ \delta_1^{k_1,k_2}\}^T$ . Assuming the hypothesis of elastic interfaces, the constitutive relation that defines interaction between block  $B_{i,j}$  and  $B_{i+k_1,j+k_2}$  is  $\sigma^{k_1,k_2} \mathbf{n}^{k_1,k_2} = \mathbf{K}^{k_1,k_2} \mathbf{d}^{k_1,k_2}$ , where  $\mathbf{n}^{k_1,k_2}$  is the vector normal to interface

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301  $S_{k_1, k_2}$ , and  $\mathbf{K}^{k_1, k_2} = \text{diag}\{K_t, K_{c2}, K_{c1}\}$  is the interface stiffness matrix, that collects  
302 tangential ( $K_t$ ) and rotational ( $K_{c1}, K_{c2}$ ) stiffness of the interface, that may be detailed  
303 for horizontal and vertical cases. Assuming mortar joints made of an isotropic and  
304 elastic material, **interface** stiffness values are function of mortar elastic modulus  $E^m$   
305 and Poisson ratio  $\nu^m$ . For instance:  
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$$307$$

$$308 \quad K_t^h = K_{c2}^h = \frac{1}{e_v} \cdot \frac{E^m}{2(1+\nu^m)} = \frac{G}{e_v}, \quad K_{c1}^h = \frac{1}{e_v} \cdot \frac{E^m}{1-(\nu^m)^2} = \frac{K}{e_v}, \quad (4a,b)$$

$$309$$

$$310 \quad K_t^v = K_{c1}^v = \frac{1}{e_v} \cdot \frac{E^m}{2(1+\nu^m)} = \frac{G}{e_v}, \quad K_{c2}^v = \frac{1}{e_v} \cdot \frac{E^m}{1-(\nu^m)^2} = \frac{K}{e_v}, \quad (5a,b)$$

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314 where  $e_h, e_v$  represent horizontal and vertical joint thickness, respectively.  
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## 316 2.1 Elastic energy

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318 The elastic energy over a generic interface is determined by defining the product of  
319 interface forces-couples and interface relative displacements-rotations:  
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$$321$$

$$322 \quad \Pi_{k_1, k_2} = \frac{1}{2} \int_{S_{k_1, k_2}} (\boldsymbol{\sigma} \mathbf{n})^T \mathbf{d} \, dS = \frac{1}{2} \int_{S_{k_1, k_2}} \mathbf{d}^T \mathbf{K} \mathbf{d} \, dS = \frac{1}{2} \mathbf{d}^T (\mathbf{K} \mathbf{A}) \mathbf{d} = \frac{1}{2} \mathbf{d}^T \bar{\mathbf{K}} \mathbf{d}, \quad (6)$$

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325 where apex  $k_1, k_2$  for vectors and matrices is omitted for simplicity,  $\mathbf{A}$  is the generic  
326 diagonal matrix of area and inertias of the interface, that may be detailed for the  
327 horizontal **case** and **for the** vertical case:  $\mathbf{A}_h = \text{diag}\{S_h, I_{h1}, (I_{h1} + I_{h3})\}$ ,  
328  $\mathbf{A}_v = \text{diag}\{S_v, (I_{v2} + I_{v3}), I_{v3}\}$ , with:  
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$$331$$

$$332 \quad S_h = b s / 2, \quad I_{h1} = b s^3 / 24, \quad I_{h3} = b^3 s / 96, \quad (7a-f)$$

$$333 \quad S_v = a s, \quad I_{v2} = a s^3 / 12, \quad I_{v3} = a^3 s / 12.$$

334  
335 **Interface** forces and couples may be obtained by differentiating the expression of  
336 **interface** elastic energy in Equation (6) with respect to each block displacement  
337 component. Then, extending such equation to the entire masonry assemblage (i.e.  
338 masonry panel), the total elastic energy  $\Pi$  is obtained and the subsequent  
339 equilibrium equation for the assemblage subject to out of plane actions  $\mathbf{F}^{ext}$  is:  
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$$342 \quad \mathbf{F}^{ext} = \partial \Pi_{panel} / \partial \mathbf{q} = \mathbf{K}^{panel} \mathbf{q}, \quad (8)$$

343  
344 where  $\mathbf{q}$  collects block degrees of freedom of the entire panel. Equation (8) may be  
345 solved by adopting a molecular dynamics algorithm or directly by explicitly  
346 defining the stiffness matrix of the entire assemblage  $\mathbf{K}^{panel}$ , extending to the out of  
347 plane case the procedure adopted by Baraldi and Cecchi [52] for the in plane case in  
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the field of modal analysis of masonry panels. Details of the determination of  $\mathbf{K}^{panel}$  may be found in appendix A.

## 2.2 Kinetic energy

The kinetic energy of a masonry assemblage may be defined at block level and it involves directly the global displacements of a generic block  $\mathbf{q}^{i,j} = \{u_3^{i,j} \ \omega_2^{i,j} \ \omega_1^{i,j}\}^T$ :

$$\Pi_{i,j}^{kin} = \frac{1}{2} [m(\dot{u}_3^{i,j})^2 + J_2(\dot{\omega}_2^{i,j})^2 + J_1(\dot{\omega}_1^{i,j})^2], \quad (9)$$

where  $m = \gamma(a \cdot b \cdot s)$  is the mass of a block having density  $\gamma$  and  $J_1 = m(a^2 + s^2)/12 = \gamma I_1$ ,  $J_2 = m(b^2 + s^2)/12 = \gamma I_2$  are block polar or rotatory inertias with respect to  $y_1$  and  $y_2$  axes [24,36]. Mass and polar inertias of each block may be corrected by taking into account mortar joint thickness, especially if it is not standard (i.e.  $e_v$  or  $e_h > 10$  mm), then  $m$  may be substituted by  $m^* = \gamma[(a + e_v) \cdot (b + e_h) \cdot s]$ ,  $J_1$  with  $J_1^* = m^*[(a + e_h)^2 + s^2]/12$  and  $J_2$  with  $J_2^* = m^*[(b + e_v)^2 + s^2]/12$ . Writing Equation (9) in matrix form, the (local) mass-polar inertia matrix  $\mathbf{M}^{i,j}$  of the generic block may be highlighted:

$$\Pi_{i,j}^{kin} = \frac{1}{2} \{\dot{u}_3^{i,j} \ \dot{\omega}_2^{i,j} \ \dot{\omega}_1^{i,j}\} \begin{bmatrix} m & 0 & 0 \\ 0 & J_2 & 0 \\ 0 & 0 & J_1 \end{bmatrix} \begin{Bmatrix} \dot{u}_3^{i,j} \\ \dot{\omega}_2^{i,j} \\ \dot{\omega}_1^{i,j} \end{Bmatrix} = \frac{1}{2} (\dot{\mathbf{q}}^{i,j})^T \mathbf{M}^{i,j} \dot{\mathbf{q}}^{i,j}, \quad (10)$$

where  $\langle \cdot \rangle = d/dt$ . The mass matrix of the entire panel  $\mathbf{M}^{panel}$  is obtained by assembling the local mass matrix  $\mathbf{M}^{i,j}$  over the panel, obtaining a diagonal mass matrix at panel level.

## 2.3 Modal analysis

The dynamic equilibrium of a masonry assemblage represented by a discrete model can be formulated by expressing the equilibrium of the effective forces associated with each of its degrees of freedom. Then, a multi degrees of freedom (MDOF) equilibrium equation is obtained [53]:

$$\mathbf{M}^{panel} \ddot{\mathbf{q}} + \mathbf{C}^{panel} \dot{\mathbf{q}} + \mathbf{K}^{panel} \mathbf{q} = \mathbf{F}^{ext}, \quad (11)$$

where  $\langle \cdot \cdot \rangle = d^2/dt^2$ ,  $\mathbf{M}^{panel}$  and  $\mathbf{K}^{panel}$  are, respectively, mass and stiffness matrices of the entire panel and  $\mathbf{C}^{panel}$  is its damping matrix (a diagonal matrix collecting damping coefficients related to block DOFs). The equation above is coincident to the one adopted by the authors for solving static problems by means of a molecular dynamics algorithm [24,54] and that is usually adopted in more general



414 discrete element modelling numerical procedures. Such equation, in general, is  
 415 solved by considering each DOF separately from other DOFs, hence it does not  
 416 request the actual determination of panel matrices. On the other hand in this work,  
 417 matrices definition is fundamental for performing static and modal analyses.  
 418 For simplicity, the equations of motion of a freely vibrating undamped system can  
 419 be obtained by omitting the damping matrix and the applied load vector:  
 420  $\mathbf{M}^{panel} \ddot{\mathbf{q}} + \mathbf{K}^{panel} \mathbf{q} = \mathbf{0}$ . Then, assuming that the free vibration motion of the panel is  
 421 simple harmonic, which may be expressed for a MDOF system as  
 422  $q_i(t) = q_i \sin(\eta t + \theta)$ , the equation of motion is modified as -  
 423  $-\eta^2 \mathbf{M}^{panel} \mathbf{q} + \mathbf{K}^{panel} \mathbf{q} = \mathbf{0}$ . Then, a standard eigenvalue problem is obtained. The  
 424 quantities  $\eta_i^2$  are the  $i$ -th eigenvalues of the vibrating system, which are related to  
 425 the free vibration frequencies of the panel  $\lambda_i = \eta_i/(2\pi)$ , while the corresponding  
 426 displacement vectors  $\mathbf{q}_i$  represent the corresponding  $i$ -th shapes of the vibrating  
 427 system, known as eigenvectors or modal shapes [53]. As well known, natural  
 428 frequencies and modal shapes are obtained by solving a standard eigenvalue  
 429 problem and finite amplitude vibrations are possible only if  
 430  $\det[\mathbf{K}^{panel} - \eta^2 \mathbf{M}^{panel}] = 0$ . Moreover, after the determination of eigenvalues and  
 431 eigenvectors of the system, the mass and polar inertia participation factor of each  
 432 eigenpair respect to out of plane translation and rotations may be determined as  
 433  $PF_i^r = (\mathbf{q}_i^T \mathbf{M}^{panel} \mathbf{r})^2 / (\mathbf{r}^T \mathbf{M}^{panel} \mathbf{r})$ , where  $\mathbf{r}$  is the vector of displacement for the  $r$ -th  
 434 degree of freedom considered. Such factors turn out to be very important for  
 435 selecting eigenpairs that activate the largest percentage of mass in  $y_3$  direction and  
 436 for evaluating the rotatory inertia activated by each vibrating mode.

### 3 Continuous models

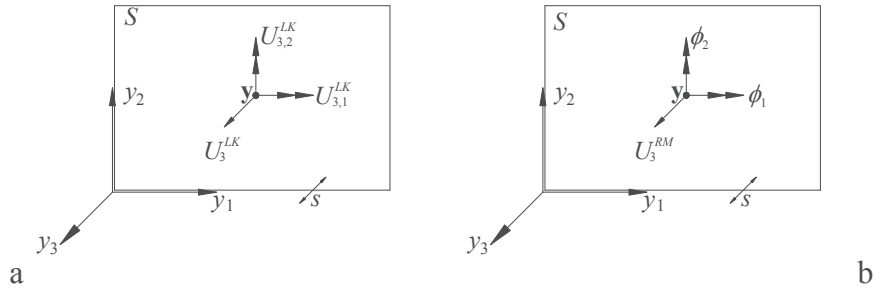


Figure 2. Love-Kirchhoff (a) and Reissner-Mindlin (b) plate models.

461 Cecchi and Sab [20,24] developed Love-Kirchhoff and Reissner-Mindlin plate  
 462 models (Figure 2 a,b respectively) for studying masonry walls, in and out of plane  
 463 loaded, by means of rigorous homogenisation procedures. It is worth noting that in  
 464 this field of analysis two scale parameters may be taken into account. Considering a  
 465 masonry assemblage with overall size  $L$ , the first scale parameter is the in plane one  
 466  $\varepsilon = b/L$  that is typical of materials with an internal structure and plane heterogeneity,  
 467

whereas the second scale parameter is the out of plane one  $\zeta = s/L$ , that is typical of plate structures and that in general does not depend on in plane heterogeneity. It is well-known that when  $\zeta$  tends to zero, the 3D solution converges to the Love-Kirchhoff solution. Caillerie [55] has extended this result to periodic plates.

### 3.1 Love-Kirchhoff plate model

Following Cecchi and Sab [24], in the Love-Kirchhoff orthotropic plate model (Figure 2a) the 3D displacement field is expressed in terms of the out of plane displacement  $U_3^{LK}(y_1, y_2)$  as follows:

$$\mathbf{u}^{LK}(\mathbf{y}) = \{-U_{3,1}^{LK}(y_1, y_2)y_3 \quad -U_{3,2}^{LK}(y_1, y_2)y_3 \quad U_3^{LK}(y_1, y_2)\}^T \quad \forall \mathbf{y}, \quad (12)$$

where  $\langle \cdot, \cdot \rangle$  indicates derivation with respect to a direction of the coordinate system adopted. An identification between the 3D DRBM and the Love-Kirchhoff plate model may be performed as

$$\mathbf{q}^{i,j} = \{u_3^{i,j} \omega_2^{i,j} \omega_1^{i,j}\}^T = \{U_3^{LK}(\mathbf{y}^{i,j}) \quad -U_{3,1}^{LK}(\mathbf{y}^{i,j}) \quad U_{3,2}^{LK}(\mathbf{y}^{i,j})\}^T. \quad (13)$$

Then, the elastic constants which relate the plate bending tensor  $\mathbf{X}$  to the curvature tensor may be highlighted as follows:

$$\mathbf{X} = \begin{Bmatrix} X_{11} \\ X_{22} \\ X_{12} \end{Bmatrix} = \begin{bmatrix} D_{1111} & D_{1122} & 0 \\ D_{1122} & D_{2222} & 0 \\ 0 & 0 & D_{1212} \end{bmatrix} \begin{Bmatrix} \chi_{11} \\ \chi_{22} \\ \chi_{12} \end{Bmatrix}, \quad (14)$$

where  $\chi_{\alpha\beta} = -U_{3,\alpha\beta}^{LK}$  with  $\alpha, \beta = 1, 2$  and  $D_{\alpha\beta\gamma\delta}$  were identified in the work of Cecchi and Sab [20]. In order to study the free vibrations of the homogenised Love-Kirchhoff plate, Equation (14) needs to be solved together with the following equilibrium equations:

$$\begin{aligned} M_{\alpha\beta,\beta} - Q_\alpha &= 0, \\ Q_{\alpha,\alpha} - \gamma s (\partial^2 U_3^{LK} / \partial t^2) &= 0, \end{aligned} \quad (15a,b)$$

where  $Q_\alpha$  with  $\alpha = 1, 2$  is the generic component of the shear stress tensor  $\mathbf{Q} = \{Q_1 \ Q_2\}^T$  and  $\gamma$  has been introduced in paragraph 2.2 for representing masonry density. Therefore, the differential equation of the Love-Kirchhoff plate is obtained and it is given by the following expression:

$$D_{1111} U_{3,1111}^{LK} + 2(D_{1122} + 2D_{1212}) U_{3,1122}^{LK} + D_{2222} U_{3,2222}^{LK} = \gamma s \frac{\partial^2 U_3^{LK}}{\partial t^2}. \quad (16)$$

The solution of the equation above is described in appendix B.1 for the determination of homogenised plate frequencies.

### 3.2 Reissner-Mindlin plate model

The Reissner-Mindlin orthotropic plate model (Figure 2b) proposed by Cecchi and Sab [24] is adopted here for taking into account shear effects in the continuous model. In this model, the 3D displacement field depends on the out of plane displacement  $U_3^{RM}(y_1, y_2)$  and two rotations  $\phi_1(y_1, y_2), \phi_2(y_1, y_2)$  as follows:

$$\mathbf{u}^{RM}(\mathbf{y}) = \{\phi_1(y_1, y_2)y_3 \quad \phi_2(y_1, y_2)y_3 \quad U_3^{RM}(y_1, y_2)\}^T \quad \forall \mathbf{y}. \quad (17)$$

Similarly to the previous model, an identification between the 3D **DRBM** and the Reissner-Mindlin plate model is defined as

$$\mathbf{q}^{i,j} = \{u_3^{i,j} \quad \omega_2^{i,j} \quad \omega_1^{i,j}\}^T = \{U_3^{RM}(\mathbf{y}^{i,j}) \quad \phi_1(\mathbf{y}^{i,j}) \quad -\phi_2(\mathbf{y}^{i,j})\}^T. \quad (18)$$

The bending elastic constants of the model are coincident to those of the Love-Kirchhoff one in Equation (14) by assuming  $\chi_{\alpha\beta} = (\phi_{\alpha,\beta} + \phi_{\beta,\alpha})/2$  with  $\alpha, \beta = 1, 2$ ; whereas the relationship between shear strains and stresses is given by

$$\mathbf{Q} = \begin{Bmatrix} Q_1 \\ Q_2 \end{Bmatrix} = \begin{bmatrix} F_{11} & 0 \\ 0 & F_{22} \end{bmatrix} \begin{Bmatrix} \gamma_{13} \\ \gamma_{23} \end{Bmatrix}, \quad (19)$$

where  $\gamma_{\alpha\beta} = \phi_\alpha + U_{3,\alpha}^{RM}$ , with  $\alpha, \beta = 1, 2$  and  $D_{\alpha\beta\gamma\delta}$  were identified by Cecchi and Sab [24]. In order to study the free vibrations of the homogenised Reissner-Mindlin plate, Equation (14) (with  $\chi_{\alpha\beta} = (\phi_{\alpha,\beta} + \phi_{\beta,\alpha})/2$ ) and Equation (19) needs to be solved together with the following equilibrium equations:

$$\begin{aligned} D_{1111}\phi_{1,11} + D_{1212}\phi_{1,22} + (D_{1122} + D_{1212})\phi_{2,12} - F_{11}(\phi_1 + U_{3,1}^{RM}) &= I_3 \frac{\partial^2 \phi_1}{\partial t^2}, \\ (D_{1122} + D_{1212})\phi_{1,12} + D_{1212}\phi_{2,11} + D_{2222}\phi_{2,22} - F_{22}(\phi_2 + U_{3,2}^{RM}) &= I_3 \frac{\partial^2 \phi_2}{\partial t^2}, \quad (20a-c) \\ F_{11}(\phi_{1,1} + U_{3,11}^{RM}) + F_{22}(\phi_{2,2} + U_{3,22}^{RM}) &= \gamma S \frac{\partial^2 U_3^{RM}}{\partial t^2}. \end{aligned}$$

Where  $I_3 = \gamma s^3 / 12$  is the rotatory inertia of the homogenised plate. The solution of the above system of equations is described in appendix B.2. It is worth noting that analytic solutions of homogenised plate models (for both plate types) in static and dynamic fields may be determined for simple load-restraint conditions such as simply supported plates. Then, in the following numerical tests, a simple

quadrilateral isoparametric shell element is adopted for representing the homogenised Reissner-Mindlin plate model if analytic solutions do not exist.

## 4 Numerical tests

A numerical experimentation is performed in order to compare different approaches that may be adopted for modelling the out of plane behaviour of masonry panels and to evaluate their field of applicability. Particular attention is given to the evaluation of the effectiveness of the homogenised models and particular attention is given to the use of discrete model for static and modal analysis. Moreover, the sensitivity of masonry out of plane behaviour to in and out of plane size of heterogeneity and to block dimension ratio is taken into account.

The following models/solution methods are adopted for performing numerical tests:

- Discrete Rigid Block Model (DRBM);
- Homogeneous Reissner-Mindlin plate model - analytic solution (RMS);
- Homogeneous Love-Kirchhoff plate model - analytic solution (LKS);
- Homogeneous Reissner-Mindlin FE model (FEM RM);
- Heterogeneous full 3D FE model (FEM Het).

DRBM and homogeneous plate models have been described in Paragraph 2; the Heterogeneous full 3D FE model is created by means of a commercial FEM code and it is characterised by 6-noded brick elements used for modelling both blocks and mortar joints by adopting different elastic parameters in order to represent the rigid block assumption (block elastic modulus  $10^4$  times larger than mortar elastic modulus). A masonry panel with a running bond texture pattern is considered as reference case for the following numerical tests (Figure 3). It is characterised by 6 blocks in horizontal direction and 16 courses in vertical direction, with blocks having dimensions  $b = 250$  mm,  $a = 55$  mm and  $s = 120$  mm. Horizontal and vertical mortar joints have the same thickness  $e = 10$  mm, hence, the overall dimensions of the panel are: length  $L = 1560$  mm, height  $H = 1040$  mm and thickness  $s = 120$  mm. The mechanical characteristics of the mortar are  $E^m = 1000$  MPa and  $\nu^m = 0.2$ .

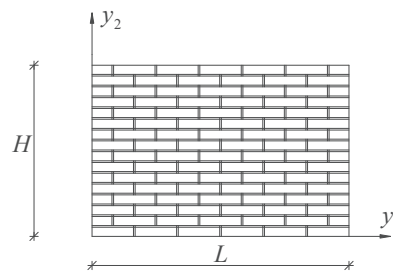


Figure 3. Masonry panel considered for the numerical examples

In the following, two panel boundary conditions are considered: panel simply supported along its edges and panel with fixed base. The first boundary condition is considered in order to provide also analytical solutions that may be determined for the homogeneous plate models considered in this work, moreover such boundary

condition may represent a masonry wall restrained along horizontal edges by slabs and along vertical edges by orthogonal walls without a perfect connection. The second boundary condition is considered in order to take into account a realistic restraint for a masonry wall with a base foundation.

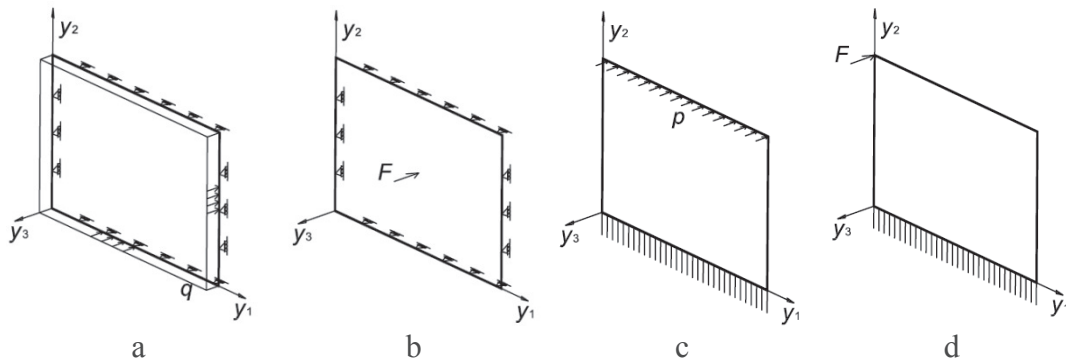


Figure 4. Case studies considered for the static analysis

#### 4.1 Static analysis

In order to perform linear static analysis, several load conditions are taken into account together with the two boundary conditions defined previously. The following list collects the case studies considered (Figure 4):

- Case study 1: panel simply supported along edges, subject to a uniform load distribution (Figure 4a);
- Case study 2: panel simply supported along edges, subject to a load distribution over a small area at its centre (Figure 4b);
- Case study 3: panel with fixed base, subject to a uniform load distribution along upper edge (Figure 4c);
- Case study 4: panel with fixed base, subject to a load distribution over a small area at its upper-left corner (Figure 4d).

All cases are obviously characterised by load distributions acting in the direction orthogonal to the middle plane of the panel. Cases 3 and 4 may represent walls in a masonry building that are perfectly connected **only** with a slab or foundation element at their base, and that are subject to horizontal forces at their top, transmitted by a well-connected roof (Case 3) or a single beam of a wooden roof (Case 4).

Before performing numerical tests related to the evaluation of scale factor effects and to the comparison of the numerical models proposed, the following figures 5 and 6 collect the deformed shapes together with the maps of out of plane displacements and rotations obtained with the reference panel for the four case studies considered, **modelled with the discrete model**. Due to the restraint condition of cases 3 and 4, out of plane rotations turn out to be negligible with respect to out of plane translations, whereas for case studies 1 and 2 all block displacement components turn out to be relevant.

#### 4.1.1 Influence of in plane size of heterogeneity – case studies 1,2

The effect of size of heterogeneity on the out of plane behaviour of masonry has been already presented in the recent work of Baraldi et al. [48]. In this work, the in plane scale factor is defined as  $\varepsilon = b/L$  (hence, reference panel in Figure 3 is characterised by  $\varepsilon = 1/6$ ) and in this paragraph block dimension ratio is maintained fixed as well as the ratio between the overall dimensions of the panel. For simplicity, an inverse expression of the scale factor  $\rho_1 = L/b = 1/\varepsilon$  is also introduced in order to represent with integers the results of the following analyses.

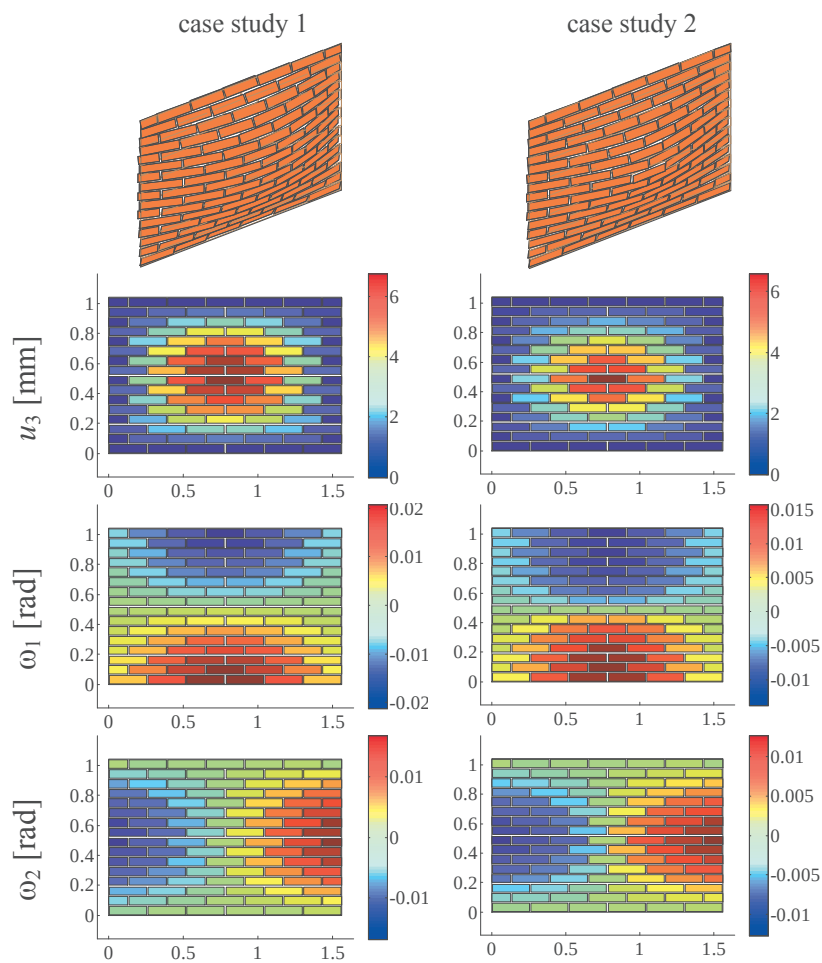


Figure 5. Out of plane displacements and rotations for case studies 1 and 2 applied to the reference panel **modelled with the discrete model**

Figure 7 shows the maps out of plane displacements obtained for case study 1, considering two different scale factors  $\varepsilon = 1/6$  and  $\varepsilon = 1/24$ , corresponding to  $\rho_1$  equal to 6 and 24, respectively, and adopting **DRBM**, Love-Kirchhoff solution (LKS) and Reissner-Mindlin solution (RMS). Following the work of Cecchi and Sab [24], analytic solutions for both Love-Kirchhoff and Reissner-Mindlin homogenised

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plate models may be determined for different load distributions over panels with simple supported edges by means of a Navier double sine series expansion. The order of magnitude of displacements is the same for the models considered, in particular **DRBM** results are close to Reissner-Mindlin solution for decreasing in plane scale factor  $\varepsilon$ . Assuming maximum displacements obtained with **DRBM** as reference solution, differences with respect to results determined with other models are evaluated:  $\text{diff} = |u_3^i - u_3^{\text{DRBM}}| / u_3^{\text{DRBM}} \cdot 100$ , where  $i$  represents a model/solution method adopted (LKS, RMS, FEM RM, FEM **Het**) and  $u_3$  is evaluated at panel midpoint. Figure 8 shows such differences for increasing  $\rho_1$  and decreasing  $\varepsilon$ , denoting that solutions determined with all models converge to **DRBM** solution and in general **FEM Het** turns out to have a behaviour very close to that of **DRBM**.

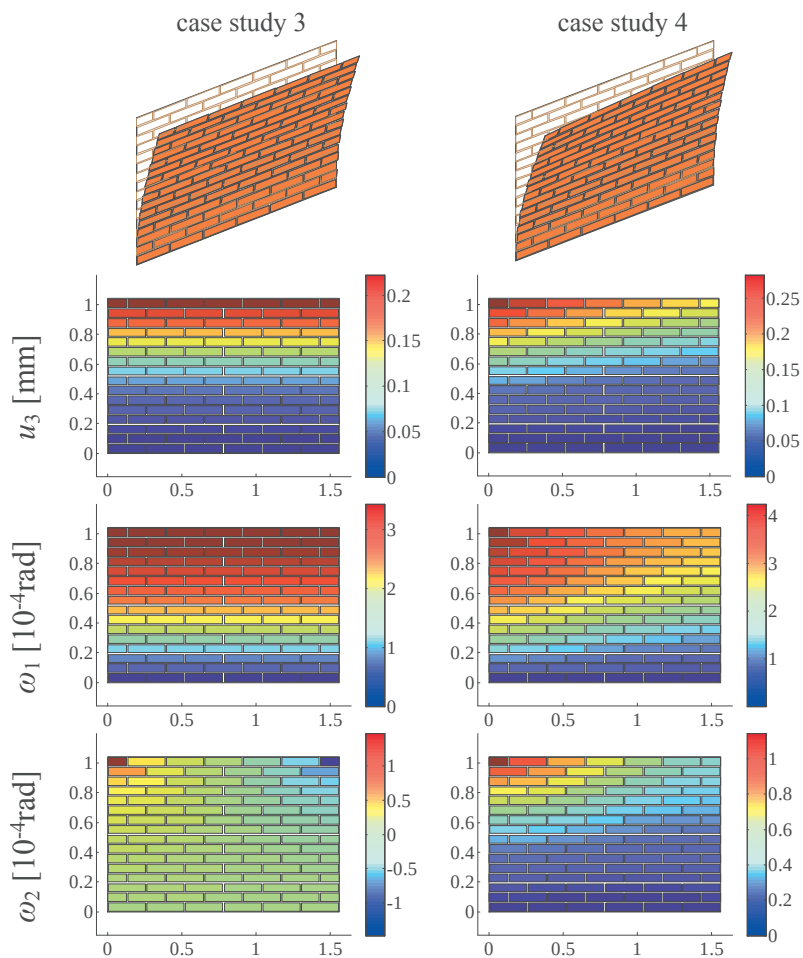


Figure 6. Out of plane displacements and rotations for case studies 3 and 4 applied to the reference panel modelled with the discrete model

For both case studies, **FEM Het** is more rigid than **DRBM**, with differences less than 5%, whereas for case study 1 homogeneous models are more deformable with respect to **DRBM** and differences are less than 5% only for  $1/\varepsilon > 15$ . Considering case study 2, all models converge to **DRBM** solution for decreasing in plane scale

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factor or increasing  $\rho_1 = 1/\varepsilon$  and in particular the Reissner-Mindlin analytic solution presents almost uniform differences with respect to **DRBM**. As can be expected, FEM RM is more deformable than the corresponding analytic solution. It is worth noting that for both case studies and  $1/\varepsilon = 3$ , all models present large differences with respect to **DRBM**; in this case panel thickness is not negligible with respect to panel size ( $s$  almost equal to  $L/6$ ), then plate models fail to represent correctly the 3D behaviour of the structure.

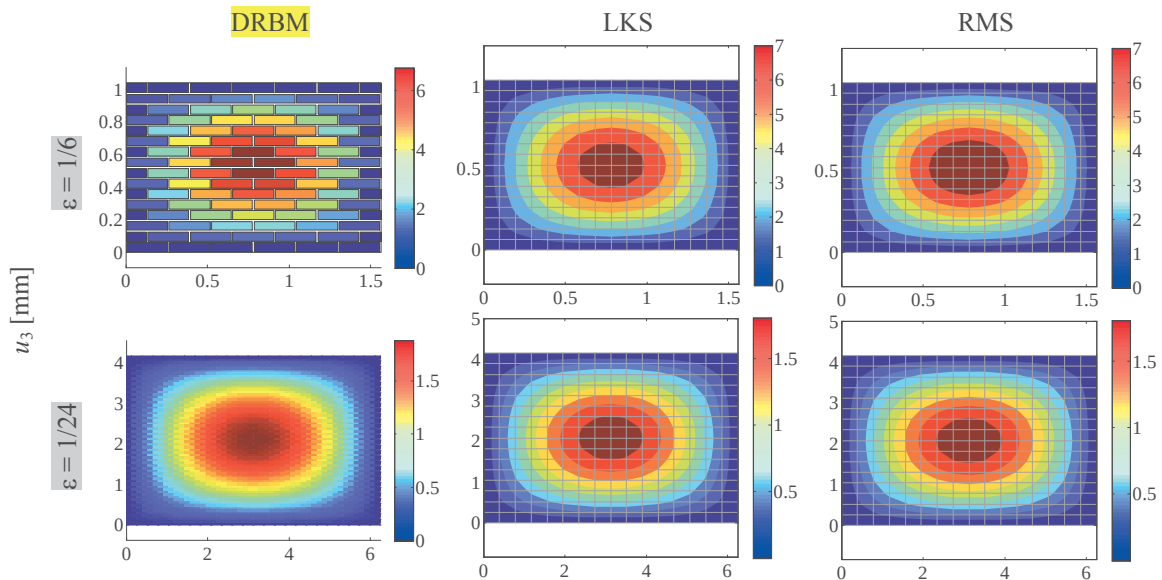


Figure 7. Out of plane displacements for case study 1

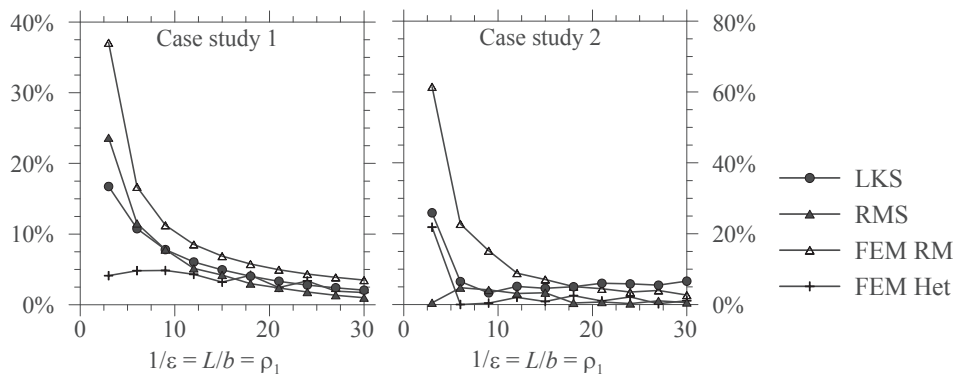


Figure 8. Differences in the determination of maximum panel displacement with respect to **DRBM** results for varying in plane scale factor

Furthermore, it is possible to compare computation times for each modelling method and for increasing in plane scale factor. Given that load and restraint conditions do not influence computation times, only case study 1 is considered for this comparison and A PC equipped with an Intel Core i7-3770 @ 3.40 GHz and 8 GB RAM is used for this purpose and analyses with 3D heterogeneous FEM are done with a rough



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mesh refinement with one element along mortar joint thickness in order to avoid huge computation times. For this reason, Figure 9 shows that computation times with the discrete model are generally less than those spent with FEM Het, but they tend to converge for increasing  $\rho_1$ . The homogeneous FE model has the advantage that the same mesh refinement, i.e. the same number of degrees of freedom, is adopted for increasing in plane scale factor, hence the computation time is constant and, adopting 32 subdivisions along both plane directions, it turns out to be smaller than that of DRBM for  $\rho_1 > 10$ .

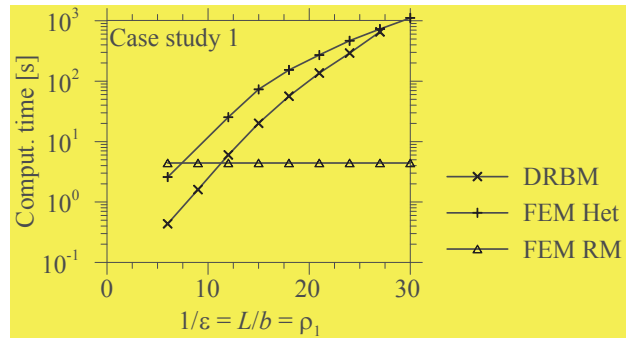


Figure 9. Computation times for varying in plane scale factor

#### 4.1.2 Influence of in plane size of heterogeneity – case studies 3,4

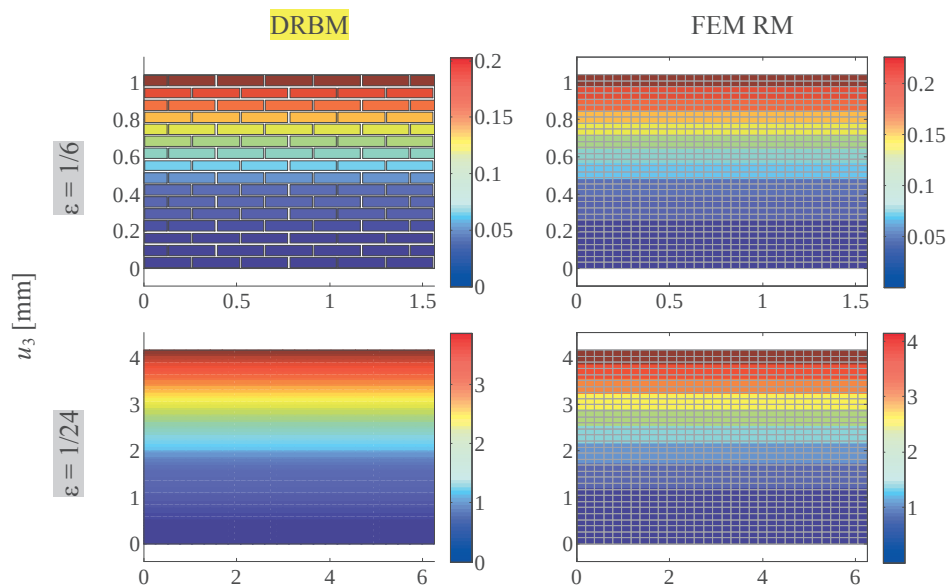


Figure 10. Out of plane displacements for case study 3

In these cases, analytic solutions for homogeneous models do not exist, then the FE model introduced for representing the behaviour of the Reissner-Mindlin plate (FEM RM) turns out to be fundamental for comparing discrete and homogenised approaches of analysis. Figures 10 and 11 show out of plane displacements obtained

for case studies 3 and 4, respectively, considering two different scale factors  $\varepsilon = 1/6$  and  $\varepsilon = 1/24$  and adopting **DRBM** and FEM RM. Similarly to the previous cases, the order of magnitude of displacements is the same for the numerical models considered, in particular for decreasing the scale factor  $\varepsilon$  or increasing  $\rho_1$ .

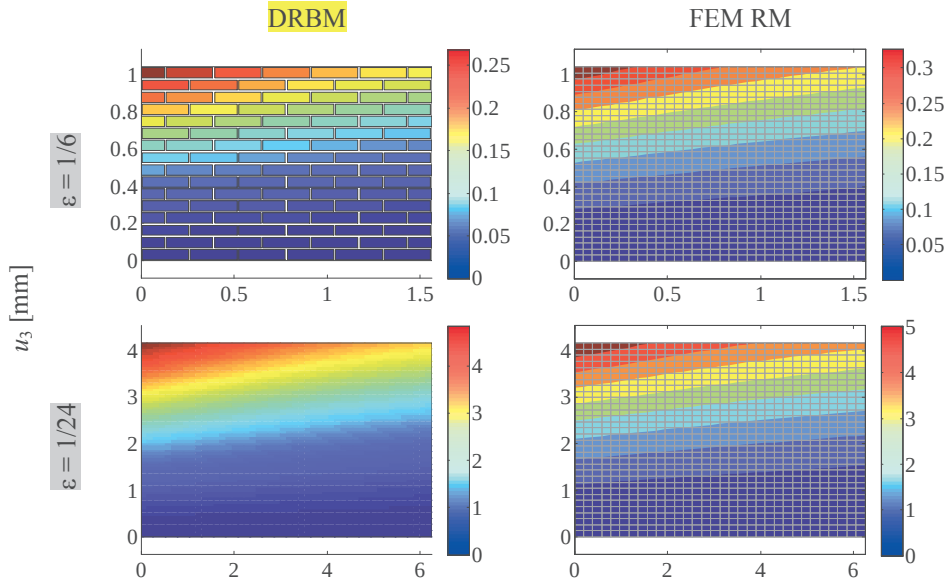


Figure 11. Out of plane displacements for case study 4

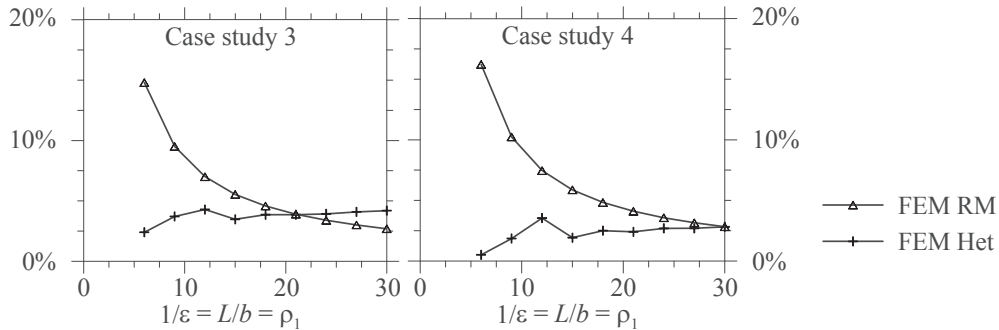


Figure 12. Differences in the determination of maximum panel displacement with respect to **DRBM** results for varying in plane scale factor

Then, assuming maximum displacements obtained with **DRBM** as reference, differences with respect to results determined with other models are evaluated for increasing  $\rho_1$  and represented in Figure 12. Similarly to the previous cases, the FEM introduced for representing the Reissner-Mindlin model is more deformable with respect to **DRBM** and differences are less than 5% only for  $\rho_1 > 15$ . The **3D heterogeneous FEM** is generally more rigid than the **DRBM** (except for  $\rho_1 = 18$  and 24) and differences are always close to 5%.

### 4.1.3 Influence of out of plane size of heterogeneity: panel thickness

In order to evaluate the effect of panel thickness on the out of plane behaviour of masonry, the out of plane scale factor  $\zeta = s/L$ , introduced previously, is here considered (for instance, the panel in Figure 3 is characterised by  $\zeta = 0.077$ ). Small values of  $\zeta$  represent thin panels, whereas large values of  $\zeta$  represent thick panels. In the following tests, a panel with  $\rho_1 = 18$  is considered and its thickness is varied assuming several values of  $\zeta$ . For simplicity only case studies 1 and 3 are taken into account.

Figure 13 shows maximum panel displacements obtained with both plate models and DRBM for increasing  $\zeta$ . Considering case study 1, it is evident that the results of both plate models, as expected, converge to DRBM results for small values of  $\zeta$ , whereas for increasing  $\zeta$ , both plate models are not able to represent the DRBM; however, as can be expected, the Reissner-Mindlin model is slightly closer to DRBM than the Love-Kirchhoff model. Considering case study 3, Figure 13 shows that differences between DRBM and FEM RM results are almost uniform for increasing  $\zeta$ , close to 5% (obtained previously in Figure 12 for  $\rho_1 = 18$ ).

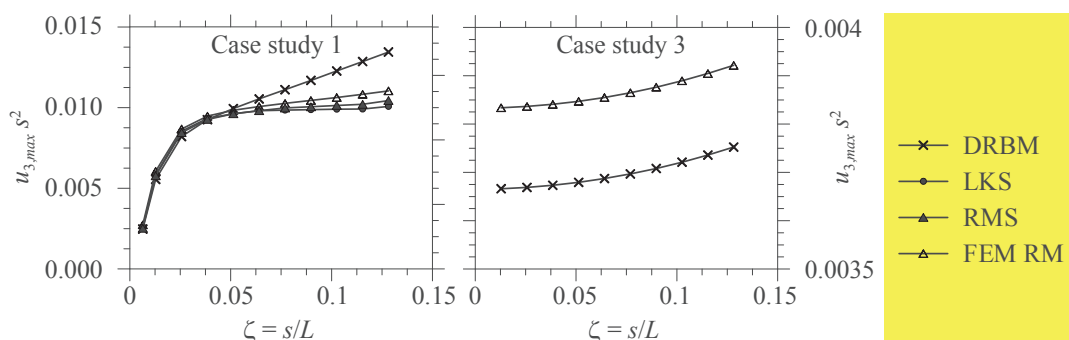


Figure 13. Maximum panel displacements for increasing out of plane scale factor

## 4.2 Modal analysis

In this paragraph, modal analysis of masonry panels modelled with DRBM is performed in order to evaluate the effectiveness of the model in the determination of out of plane panel vibration frequencies and of the corresponding modal shapes and in order to evaluate the effects of in and out of plane scale factors, together with block and panel dimension ratios.

### 4.2.1 Panel with simply supported edges

The panel with simply supported edges adopted for case studies 1 and 2 (Figure 4 a,b) is taken into account for first. Figure 14 shows the first four mode shapes and the corresponding frequencies for the reference panel; block thickness is not represented for simplicity and the corresponding colour maps of out of plane displacement are added for better understanding block displacements. Similarly to isotropic plates, first mode shape is characterised by one half-wave in both plane

directions, second mode shape has one half wave in  $y_1$  direction and 2 half waves in  $y_2$  direction, third mode shape has 2 half waves in  $y_1$  direction and one half-wave in  $y_2$  direction, fourth mode shape has 2 half-waves in both plane directions.

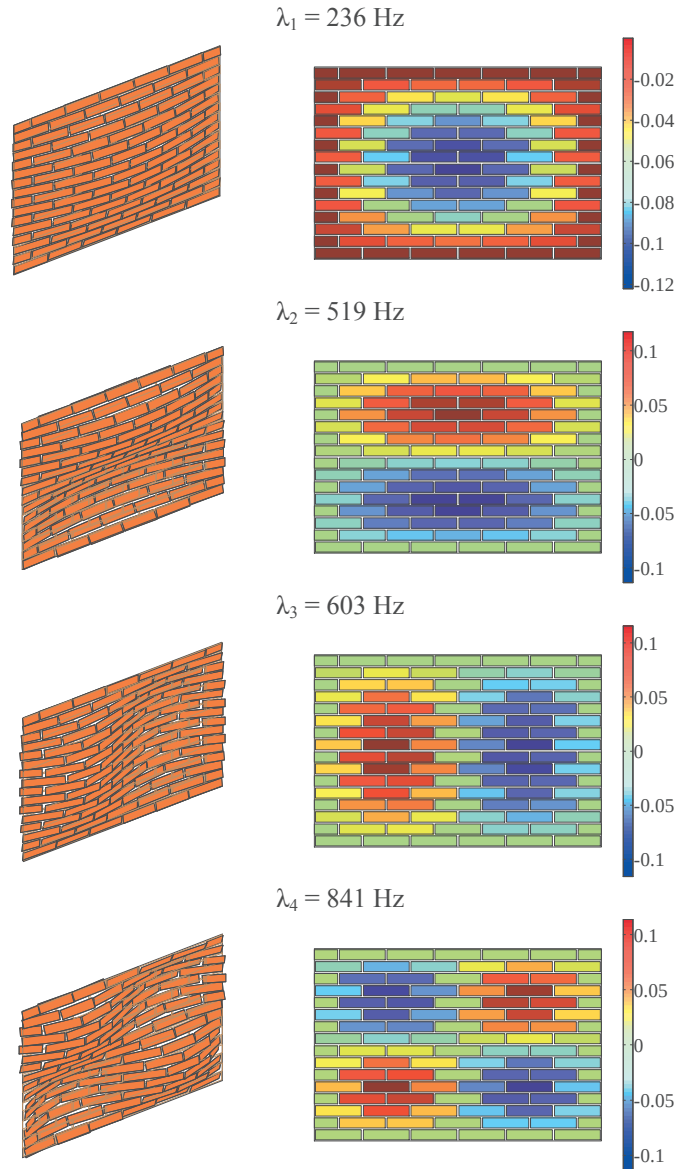


Figure 14. First four mode shapes, maps of out of plane displacements and corresponding frequencies for a simply supported masonry panel taken as reference (Figure 3) and modelled with the discrete model

Analytic solutions for both Love-Kirchhoff and Reissner-Mindlin homogeneous plates may be determined by solving the corresponding equation of motion (Eqs. 16 and 20 a-c, respectively). In appendix B, analytic expressions for frequencies are determined, in order to compare the corresponding results with respect to DRBM results. Figure 15 shows differences  $\text{diff}_i^j = (\lambda_i^j - \lambda_i^{\text{DRBM}}) / \lambda_i^{\text{DRBM}} \cdot 100$  in the

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determination of the first four frequencies assuming **DRBM** results as references, where  $j$  indicates a homogeneous plate solution method or solutions obtained with heterogeneous FEM and  $i$  indicates the  $i$ -th eigenpair considered. Frequencies obtained with LKS and RMS appear to be quite close to each other and present similar differences with respect to **DRBM** results. Hence for both analytical models, differences obtained for 1st and 4th frequencies decrease to 2% for decreasing in plane scale factor  $\varepsilon$  or increasing  $\rho_1$ , whereas differences obtained for 2nd and 3rd frequencies slightly decrease for increasing  $\rho_1$  and are close to 10%. Similarly to the static case, the results obtained with the **3D** heterogeneous FEM are closer to **DRBM** results with respect to homogeneous plate solutions.

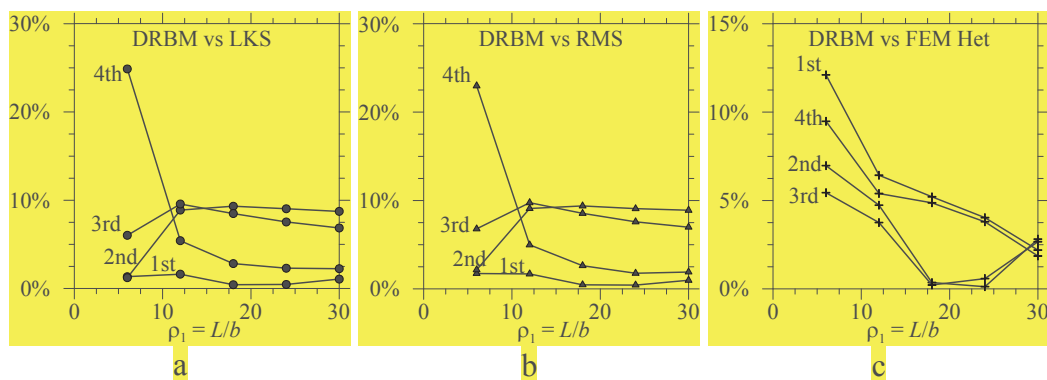


Figure 15. Differences in the determination of panel frequencies with respect to **DRBM** results for increasing  $\rho_1$  or decreasing in plane scale factor  $\varepsilon$

#### 4.2.2 Panel with fixed base

In this paragraph, the panel with fixed base adopted for case studies 3 and 4 (Figure 4 c,d) modelled with **DRBM** is taken into account. Panel texture in Figure 3 is assumed as reference case, then several geometrical parameters are modified in order to evaluate their effect on panel frequencies and modal shapes. For the following analyses a second in plane ratio  $\rho_2 = H/a$  is introduced in order to evaluate block dimension ratio  $a/b$ .

Then, the following aspects are considered for performing modal analysis:

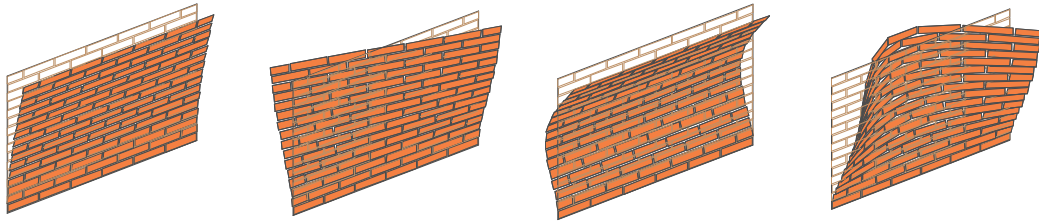
- sensitivity to in plane scale factor: maintaining fixed block dimensions, the number of blocks in both plane directions is increased;
- sensitivity to out of plane scale factor: maintaining fixed block and panel dimensions (reference case), the effect of varying panel thickness is taken into account;
- sensitivity to block dimension ratio - subcase 1: maintaining fixed panel dimensions and block height  $a$ , the effect of varying block width  $b$  is taken into account;
- sensitivity to block shape factor - subcase 2: maintaining fixed panel dimensions and block width  $b$ , the effect of varying block height  $a$  is taken into account;
- sensitivity to panel shape factor: maintaining fixed block dimensions and panel width  $L$ , the effect of varying panel height  $H$  is taken into account.

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 1187 The first row of Figure 16 shows the first four mode shapes of the reference case  
 1188 together with the corresponding frequencies. The first mode shape is characterised  
 1189 by the classical flexural deformation in vertical direction with one half-wave and  
 1190 almost 60% of participant mass in out of plane direction ( $PF_1^{u_3} = 59.3\%$ ). The  
 1191 second mode shape is characterised by a torsional deformation; in this case the  
 1192 participant mass in out of plane direction is close to zero, whereas a participant  
 1193 rotatory inertia with respect to vertical direction ( $PF_2^{\omega} = 2.2\%$ ) is obtained. The  
 1194 third mode shape is characterised by a flexural deformation with two half-waves and  
 1195 a  $PF_3^{u_3} = 18.6\%$ . The fourth mode shape is characterised by flexural deformations in  
 1196 both plane directions (flex. 2D) but mass and rotatory inertia participation factors are  
 1197 close to zero.

1200 In order to evaluate the effect of in plane scale factors, the second row of Figure 16  
 1201 shows the first four mode shapes and the corresponding frequencies of a panel  
 1202 having  $\varepsilon = 1/24$ , corresponding to  $\rho_1 = 24$  and  $\rho_2 = 64$ . Mode shapes order does not  
 1203 vary with respect to the reference case, whereas frequencies are smaller than those  
 1204 of the reference case due to the larger deformability of the panel given by the large  
 1205 number of blocks and joints considered. Mass participation factors in out of plane  
 1206 direction of first and third mode shapes are almost coincident to those of the  
 1207 reference case ( $PF_1^{u_3} = 60.1\%$ ,  $PF_3^{u_3} = 18.6\%$ ). Then, in plane scale factor does not  
 1208 affect significantly modal shapes and mass participation factors of fixed base panels.  
 1209 Figure 17a, indeed, shows that frequencies decrease linearly and mode order do not  
 1210 vary for increasing  $\rho_1$  or decreasing  $\varepsilon$ .

1212 In order to evaluate the effect of out of plane scale factor, Figure 17b shows the first  
 1213 four frequencies for increasing out of plane factor  $\zeta$ . As can be expected, frequencies  
 1214 increase linearly for increasing  $\zeta$  due to the increasing stiffness of the panel given by  
 1215 its thickness, moreover mode order does not vary for increasing  $\zeta$  with respect to the  
 1216 reference case and for this reason deformed shapes are not added to Figure 16. In  
 1217 order to evaluate block dimension ratio  $a/b$ , the third row of Figure 16 shows the  
 1218 first four mode shapes and the corresponding frequencies for a panel having  $\rho_1 = 24$   
 1219 and  $\rho_2 = 16$ , corresponding to  $a/b = 1.0$ . It is clear that the first mode shape and the  
 1220 corresponding frequency is quite close to the first eigenpair of the reference case,  
 1221 however the subsequent mode shapes are characterised by different order and  
 1222 frequency values with respect to the reference case. Figure 17c, indeed, shows that  
 1223 the first frequency value does not vary significantly for increasing  $\rho_1$  and similarly,  
 1224 the frequency values corresponding to the flexural mode shape with two half-waves  
 1225 are almost constant or slightly increase for increasing  $\rho_1$ . On the other hand,  
 1226 frequencies corresponding to torsional mode shape and the one with flexure in both  
 1227 plane directions decrease for increasing  $\rho_1$  due to the increasing deformability given  
 1228 by the increasing number of blocks and mortar joints in horizontal direction. Hence,  
 1229 it is clear that flexural mode shapes are not affected significantly by scale factor  $\rho_1$  if  
 1230 block height and panel dimensions are fixed. Continuing to consider the effect of  
 1231 block dimension ratio  $a/b$ , the fourth row of Figure 16 shows the first four mode  
 1232 shapes of a panel having  $\rho_1 = 6$  and  $\rho_2 = 32$ , corresponding to  $a/b = 0.125$ .

Reference case



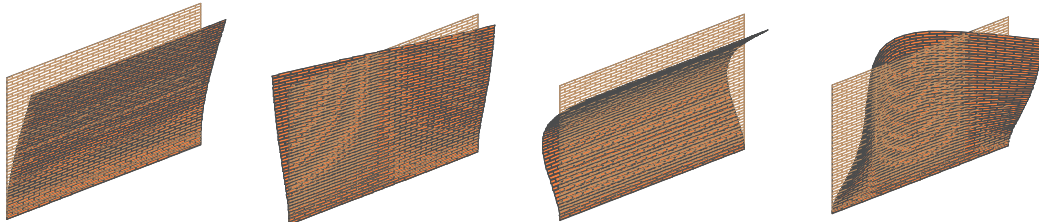
$\lambda_1 = 40.7$  Hz

$\lambda_2 = 116.3$  Hz

$\lambda_3 = 242.7$  Hz

$\lambda_4 = 338.2$  Hz

Sensitivity to in plane scale factor



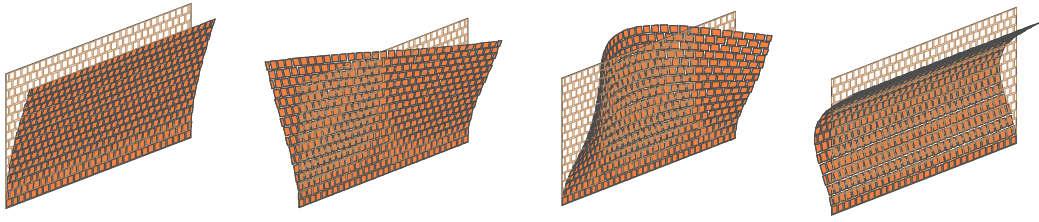
$\lambda_1 = 2.4$  Hz

$\lambda_2 = 7.8$  Hz

$\lambda_3 = 15.3$  Hz

$\lambda_4 = 23.2$  Hz

Sensitivity to block shape factor - subcase 1



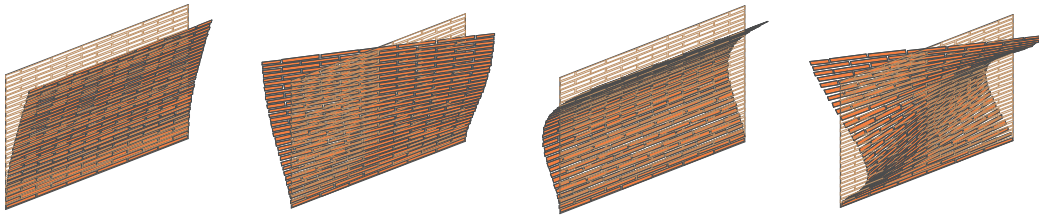
$\lambda_1 = 43.4$  Hz

$\lambda_2 = 72.0$  Hz

$\lambda_3 = 164.8$  Hz

$\lambda_4 = 258.8$  Hz

Sensitivity to block shape factor - subcase 2



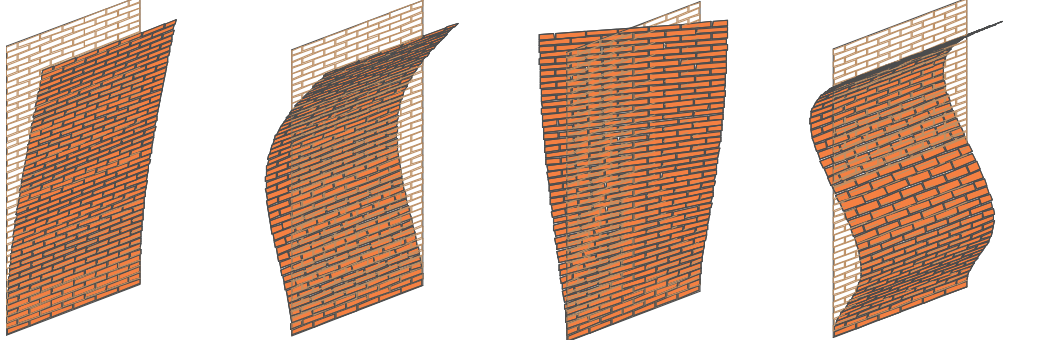
$\lambda_1 = 30.8$  Hz

$\lambda_2 = 127.5$  Hz

$\lambda_3 = 15.3$  Hz

$\lambda_4 = 23.2$  Hz

Sensitivity to panel shape factor



$\lambda_1 = 4.4$  Hz

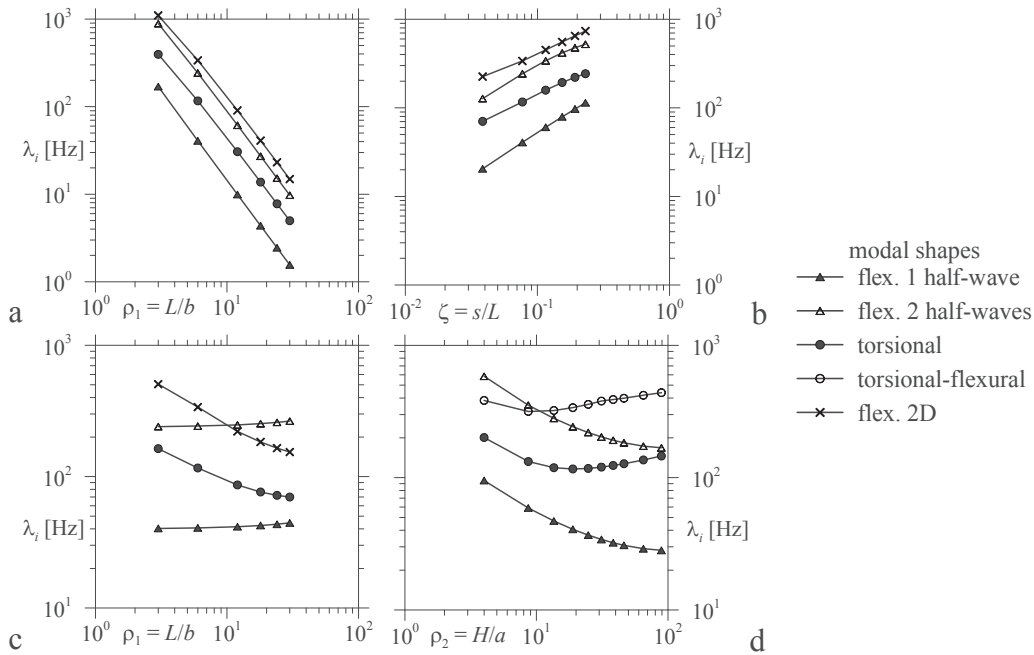
$\lambda_2 = 27.2$  Hz

$\lambda_3 = 33.4$  Hz

$\lambda_4 = 75.5$  Hz

Figure 16. Modal analysis of masonry panels (DRBM) varying several parameters

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 1304 First and third mode shapes are characterised by flexural deformation with one and  
 1305 two half-waves, respectively, whereas second mode shape is a simple torsional  
 1306 deformation and the fourth mode shape is a combined flexural-torsional  
 1307 deformation. Frequencies corresponding to flexural modal shapes decrease for  
 1308 increasing  $\rho_2$  (Figure 17d), whereas frequencies corresponding to torsional modal  
 1309 shapes initially decrease up to  $\rho_2 = 10$  and then increase for  $\rho_2 > 10$ .  
 1310



1331 Figure 17. First four panel frequencies varying in plane and out of plane scale  
 1332 factors. Sensitivity to: in plane scale factor (a), out of plane scale factor (b); block  
 1333 shape factor - subcase 1 (c), block shape factor - subcase 2 (d)  
 1334

1335 Finally, the effect of panel shape factor is considered by varying  $L/H$  ratio. For  
 1336 example, the fifth row of Figure 16 collects the first four modal shapes  
 1337 corresponding to  $L/H = 0.5$ , with  $\rho_1 = 6$ . Thanks to the slenderness of the panel, first,  
 1338 second and fourth mode shapes are flexural with one, two and three half-waves,  
 1339 respectively, whereas the third modal shape is torsional. Figure 18a shows frequency  
 1340 values for increasing  $L/H$ . For the range of  $L/H$  values considered, the first mode  
 1341 shape is always flexural with one half-wave and it is possible to define an expression  
 1342 for estimating frequency values linearly depending on  $L/H$  and panel material  
 1343 parameters. Moreover, Figure 18b shows mass participation factors related to this  
 1344 case study and the first mode shape tends to activate the 60% of the total panel mass  
 1345 for decreasing  $L/H$ , whereas the percentage of mass activated by the second flexural  
 1346 mode shape does not depend on  $L/H$  and it is close to 20%. Figure 18c shows  
 1347 rotatory inertia participation factors, that are strictly related to block out of plane  
 1348 rotations. In particular, the rotatory inertia related to  $\omega_2$  for the first and second  
 1349 flexural modes increases significantly for increasing  $L/H$ .  
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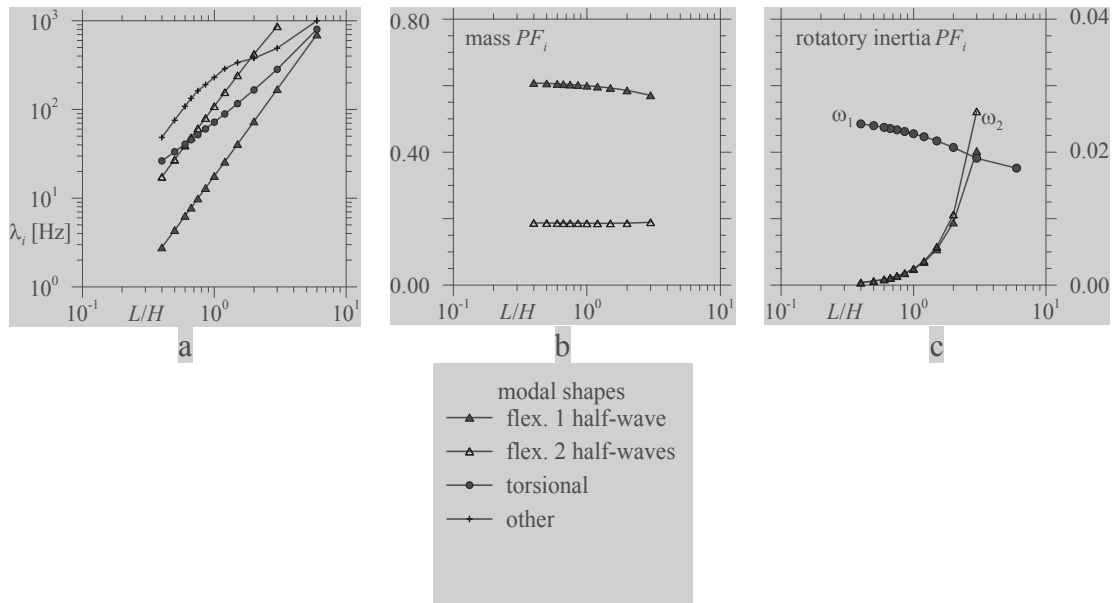


Figure 18. Modal analysis varying panel shape factor  $L/H$ . Frequency values (a), mass participation factors (b); rotatory inertia participation factors (c)

## 4 Conclusions

In this work the out of plane behaviour of masonry panels with regular texture has been considered. Elastic behaviour, small displacements, and fixed contact topology are the hypotheses adopted for the analysis. Several numerical and analytical models have been taken into account for modelling masonry, with particular attention to a simple discrete model introduced by Cecchi and Sab [24] and here extended by performing static analysis with several boundary conditions and load cases and, in particular, by performing out of plane modal analysis. The proposed review has shown that the discrete model (DRBM) is simple, effective and efficient for modelling out of plane behaviour, thanks to the small number of degrees of freedom involved during the analyses. Such simplicity allowed to define the stiffness matrix of the regular assemblage of blocks, in order to obtain fast static solutions with respect to the molecular dynamics method adopted originally and in order to perform modal analysis by solving eigenvalue problems.

Both in static and modal analysis, the 3D Heterogeneous FEM turned out to be quite close to the DRBM, however it is affected by the large number of degrees of freedom needed for modelling accurately the masonry structure and by the mesh refinement required for obtaining accurate results. Love-Kirchhoff and Reissner-Mindlin homogeneous plate models turned out to be more deformable than the DRBM almost in all case studies considered in static and modal analysis. However, performing static analysis, analytic solutions obtained with the Reissner-Mindlin plate turned out to be closer to DRBM for decreasing in plane scale factor. In general, for the four case studies considered during static analysis tests, the

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1422 homogeneous models proved to be effective for representing masonry behaviour  
1423 only for small values of the in plane scale factor (i.e.  $\varepsilon < 1/15$  or  $\rho_1 > 15$ ).

1424 The FEM adopted for modelling the Reissner-Mindlin homogeneous plate for static  
1425 analysis turned out to be slightly more deformable than the corresponding analytical  
1426 model, however it has been fundamental in the determination of out of plane  
1427 displacements for the case studies of panels with fixed base. Furthermore,  
1428 computation times with FEM RM are not affected by the in plane scale factor, hence  
1429 this model may be successfully adopted in case of panels with a large number of  
1430 blocks.

1431 Then, modal analysis performed for the simply supported plate allowed authors to  
1432 define the analytic solutions in terms of frequencies for Love-Kirchhoff and  
1433 Reissner-Mindlin homogeneous plate models. Similarly to the static analysis case,  
1434 homogeneous models turned out to be effective for representing masonry behaviour  
1435 only for small values of the in plane scale factor (i.e.  $\varepsilon < 1/15$  or  $\rho_1 > 15$ ). Further  
1436 modal analyses on masonry panels with fixed base have been carried on adopting  
1437 only discrete model, obtaining flexural deformation with one half-wave as first  
1438 mode shape and obtaining often torsional deformation as second mode shape. In  
1439 plane and out of plane scale factors turned out to slightly influence mode shapes  
1440 order.

1441  
1442 Further developments of this work will regard the extension of the DRBM to the  
1443 field of nonlinear analysis by assuming rigid block and nonlinear interfaces,  
1444 similarly to recent developments performed by authors in plane case [50].  
1445 Furthermore, more accurate continuous models, such as the Cosserat continuum,  
1446 will be taken into account and compared to the DRBM here reviewed.

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1449  
1450  
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1453 materials and structures 664 for the solution of 2020 Horizon challenges”).

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## Appendix

### A Stiffness matrix of a regular block assemblage

In order to describe how to determine the stiffness matrix of a regular masonry assemblage, the simplest procedure envisages the definition of the stiffness matrix for an interface  $S_{k_1, k_2}$  between two generic blocks. Then, the degrees of freedom related to the generic block  $B_{i,j}$  in Equation (1) have to be taken into account together with the degrees of freedom of a generic neighbour  $B_{i+k_1, j+k_2}$ :

$$\mathbf{q}^{i+k_1, j+k_2} = \{u_3^{i+k_1, j+k_2} \ \omega_1^{i+k_1, j+k_2} \ \omega_2^{i+k_1, j+k_2}\}^T, \quad (21)$$

where  $k_1, k_2 = \pm 1$ , for horizontal interfaces,  $k_1 = \pm 2$  and  $k_2 = 0$  for vertical interfaces. The degrees of freedom of the couple of adjacent blocks may be collected in the following vector having 6 components:

$$\mathbf{q}^{k_1, k_2} = \{u_3^{i,j} \ \omega_1^{i,j} \ \omega_2^{i,j} \ u_3^{i+k_1, j+k_2} \ \omega_1^{i+k_1, j+k_2} \ \omega_2^{i+k_1, j+k_2}\}^T. \quad (22)$$

Adopting the notation of vectors  $\mathbf{q}^{i,j}$  and  $\mathbf{q}^{i+k_1, j+k_2}$ , Equation (3a-c), representing relative displacements between the blocks, may be written in matrix form as follows:

$$\begin{aligned} \mathbf{d}^{k_1, k_2} = \begin{Bmatrix} d_3^{k_1, k_2} \\ \delta_1^{k_1, k_2} \\ \delta_2^{k_1, k_2} \end{Bmatrix} = \begin{bmatrix} 1 & -k_2 a / 2 & k_1 b / 4 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} u_3^{i+k_1, j+k_2} \\ \omega_1^{i+k_1, j+k_2} \\ \omega_2^{i+k_1, j+k_2} \end{Bmatrix} + \\ - \begin{bmatrix} 1 & k_2 a / 2 & -k_1 b / 4 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} u_3^{i,j} \\ \omega_1^{i,j} \\ \omega_2^{i,j} \end{Bmatrix} = \mathbf{H}^{i+k_1, j+k_2} \mathbf{q}^{i+k_1, j+k_2} - \mathbf{H}^{i,j} \mathbf{q}^{i,j}, \end{aligned} \quad (23)$$

and, assembling matrix and vector components, the following expression may be obtained:

$$\mathbf{d}^{k_1, k_2} = \begin{bmatrix} -\mathbf{H}^{i,j} & 0 \\ 0 & \mathbf{H}^{i+k_1, j+k_2} \end{bmatrix} \begin{Bmatrix} \mathbf{q}^{i,j} \\ \mathbf{q}^{i+k_1, j+k_2} \end{Bmatrix} = \mathbf{H}^{k_1, k_2} \mathbf{q}^{k_1, k_2}. \quad (24)$$

It is worth noting that  $k_1$  and  $k_2$  depend on the interface considered, consequently matrices  $\mathbf{H}^{i+k_1, j+k_2}$ ,  $\mathbf{H}^{i,j}$  and then matrix  $\mathbf{H}^{k_1, k_2}$  depend on the interface considered. Moreover, interface forces and couples depend on relative displacements between blocks by means of constitutive relations involving interface stiffness matrices. Then, substituting the equations above in the expression of the elastic energy of the